



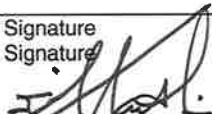

**ADDITIONAL WORK TO BE PERFORMED UPON
ENGINE INSTALLATION**

TRANSPORT CANADA
A.M.O. APPROVAL 1-58 /
EASA.145.7052

ENGINE MARK: TAY 611-8	ENGINE SERIAL NO. : 16844	SALES ORDER: 1365438
TSN: 5803.2	CSN: 2302	

THE FOLLOWING SERVICE BULLETINS MARKED WITH AN "X" MUST BE COMPLETED ON ENGINE INSTALLATION. A LOGBOOK ENTRY IS REQUIRED TO INDICATE WHICH ORGANIZATION OR INDIVIDUAL HAS COMPLETED THE WORK LISTED BELOW, THIS ENTRY MUST BE DATED AND STAMPED.	
<input checked="" type="checkbox"/>	SB 71-1147 FOUND EMBODIED IN PART, ALL RELEVANT PARTS MUST BE COMPLIED WITH PRIOR TO ENGINE INSTALLATION AS DETAILED IN ROLLS-ROYCE SERVICE BULLETIN 71-1147.
<input checked="" type="checkbox"/>	SB 76-1130 MUST BE COMPLIED WITH PRIOR TO ENGINE INSTALLATION AS DETAILED IN THE ROLLS-ROYCE SERVICE BULLETIN.
<input checked="" type="checkbox"/>	SB 71-1100 MUST BE COMPLIED WITH PRIOR TO ENGINE INSTALLATION AS DETAILED IN THE ROLLS-ROYCE SERVICE BULLETIN
<input checked="" type="checkbox"/>	SB 73-1354 TO BE COMPLIED WITH DURING ENGINE GROUND RUNS AS DETAILED IN THE ROLLS-ROYCE SERVICE BULLETIN. THIS SB MUST BE COMPLIED WITH ON BOTH ENGINES INSTALLED.

SIGNATURE: <i>Thomas Pagei</i>	Thomas Pagei	DATE: NOV 20 2018	STAMP: 
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1. Approving Civil Aviation Authority/Country Autorité de l'aviation civile/Pays qui approuve le bon de sortie Transport Canada		2. AUTHORIZED RELEASE CERTIFICATE BON DE SORTIE AUTORISÉE FORM ONE			3. Form Tracking No./Numéro de suivi du formulaire 2018-6000057	
4. Organization Name & Address Nom et adresse de l'organisme Rolls-Royce Canada Limitée 9500, Cote de Liesse, Lachine Québec, Canada H8T 1A2				5. Work Order/Contract/Invoice Bon de travail/ Contrat/Facture Customer: ALBINATI AERONAUTICS S.A. Reference P.O.9000617083 Sales Order: 1365438		
6. Item Article	7. Description Description	8. Part No. Numéro de pièce	9. Qty. Qtée.	10. Serial/Batch No. Numéro de série/de lot	11. Status/Work Situation/Travail	
01	TAY AERO ENGINE	TAY 611-8	1	16844	Overhauled	
12. Remarks/Remarques REFER TO CERTIFICATE OF COMPLIANCE FOR PERTINENT DETAILS OF WORK PERFORMED DURING THIS SHOP VISIT FILED AT THIS ORGANIZATION UNDER SALES ORDER: 1365438. THE ENGINE WAS RETURNED FROM SERVICE DUE TO FULL OVERHAUL AND WAS OVERHAULED, MODIFIED AND TESTED IN ACCORDANCE WITH ENGINE MANUAL E-TAY-1RR REV.108, ROLLS-ROYCE NMSB70-1050 REV.9, APPENDIX 6, CHAPTER 5 OF THE MAINTENANCE MANUAL M-TAY-1RR REV.135 AND CUSTOMER P.O. 9000617083. ADDITIONNAL WORK TO BE PERFORMED UPON ENGINE INSTALLATION: SB71-1100 AND 76-1130 TO BE FULLY EMBODIED. SB73-1354 TO BE COMPLIED WITH DURING ENGINE GROUND RUNS. NOTE: ENGINE FUEL SYSTEM HAS BEEN FULLY INHIBITED. TSN: 5803.2 HOURS CSN: 2302 CYCLES TSO:NIL CSO:NIL						
13a. Certifies that the items identified above were manufactured in conformity to: Le présent bon certifie que les articles indiqués ci-dessus ont été construits conformément à: <input type="checkbox"/> Approved design data and are in condition for safe operation Des données de conception approuvées et qu'ils peuvent être utilisés en toute sécurité <input type="checkbox"/> Non approved design data specified in block 12. Des données de conception non approuvées indiquées à la case 12.				14a. <input checked="" type="checkbox"/> CAR 571.10 (Maintenance release) / RAC 571.10 (Certification après maintenance). <input type="checkbox"/> Other regulation specified in block 12. Autre réglementation précisée à la case 12. Certifies that unless otherwise specified in block 12, the work identified in block 11 and described in block 12 has been performed in compliance with the <i>Canadian Aviation Regulations</i> . Le présent bon de sortie certifie que, sauf indication contraire à la case 12, le travail indiqué à la case 11 et décrit à la case 12, a été effectué conformément au <i>Règlement de l'aviation canadien</i> .		
13b. Signature Signature		13c. Approved Organization Number Numéro de l'organisme agréé		14b. Signature Signature  		14c. Approved Organization Number Numéro de l'organisme agréé AMO 1-58
13d. Name Nom		13e. Date (dd/mmm/yyyy) Date (jj/mmm/aaaa)		14d. Name Nom JEAN-MARC GAUTHIER		14e. Date (dd/mmm/yyyy) Date (jj/mmm/aaaa) 17/Nov/2018

(Previously form 24-0078) Installer Responsibilities / (Ancien formulaire 24-0078) Responsabilités du monteur.

This certificate does not constitute authority to install.
Installer working in accordance with the national regulations of a country other than that specified in block 1 must ensure that their regulations recognize certifications from the country specified.
Statements in blocks 13a or 14a do not constitute installation certification. In all cases, the technical record for the aircraft must contain an installation certification issued in accordance with the applicable national regulations before the aircraft may be flown.

Le présent bon de sortie ne constitue pas une autorisation de montage.
Le monteur qui travaille conformément à la réglementation d'un pays autre que celui spécifié à la case 1 doit s'assurer que la réglementation en question reconnaît la certification du pays ainsi spécifié.
Les déclarations des cases 13a et 14a ne constitue pas une certification de montage. Dans tous les cas, le dossier technique de l'aéronef doit inclure une certification de montage délivrée conformément à la réglementation nationale qui s'applique, avant que l'aéronef puisse voler.



Rolls-Royce

Rolls-Royce Canada Limitée
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Québec, Canada, H8T 1A2

AMO 1-58

CERTIFICATE OF COMPLIANCE

CERTIFICAT de CONFORMITÉ

<u>Make - Constructeur</u>	<u>Model- Modèle</u>	<u>Serial No. - No de Série:</u>	<u>Sales Order - Commande client</u>
ROLLS-ROYCE	Tay 611-8 G-IV	16844	1365438
TSN: 5803.2		TSO: NIL	
CSN: 2302		CSO: NIL	

The following MODs / Repairs have been accomplished this shop visit:

MODIFICATIONS EMBODIED:

- SB72-1060 REV0 H.P. TURBINE CASE AND ROTOR - INCREASED CLEARANCE BETWEEN H.P. STAGE 1 ROTOR BLADES AND NOZZLE VANES
- SB72-1252 REV0 I.P. COMPRESSOR AIR SPLITTER AND INLET GUIDE VANES - SPLITTER FAIRING AND STATOR RETAINING RINGS WITH IMPROVED CORROSION PROTECTION
- SB72-1432 REV0 ENGINE - LOW SPEED GEARBOX DRIVEN GEAR ASSEMBLY !! INTRODUCTION OF REVISED DRIVEN SHAFT NUT
- SB72-1472 REV0 ENGINE - GENERATOR DRIVE SHAFT AND BEVEL GEAR SHAFT ASSEMBLIES - HIGH SPEED GEARBOX DEE HEAD BOLT MATERIAL AND TORQUE TIGHTENING SETTING CHANGE
- SB72-1480 REV0 ENGINE HIGH SPEED GEAR BOX DRIVING ASSEMBLY INTRODUCTION OF INCREASED TORQUE ON THE INTERNAL GEARBOX BEVEL GEAR RETAINING NUT
- SB72-1499 REV0 ENGINE - HIGH SPEED GEARBOX DRIVING GEAR ASSEMBLY - INTERNAL GEARBOX - INTERFERENCE FIT OF DRIVE SHAFT AND BEVEL GEAR
- SB72-1589 REV0 ENGINE - TURBINE ROTOR DISCS AND SHAFT-LP - INTRODUCTION OF DECREASED THICKNESS LP TURBINE STAGE 1 AND 2 RETAINING AND LOCKING PLATES
- SB72-1590 REV0 TURBINE NOZZLE CASE AND VANES-LP - DELETION OF THE DOWEL PINS AND RETAINER PLATES WHICH ARE INSTALLED IN THE LP TURBINE STAGE 1 STATIC AIR SEAL
- SB72-1704 REV0 AIR GUIDE TUBE DRIVE SHAFT AND FITTINGS - HIGH PRESSURE COMPRESSOR - INTRODUCTION OF REVISED TORQUE TIGHTENING PROCEDURE FOR THE LOW PRESSURE (LP) THRUST BEARING OUTER RING SPANNER NUT
- SB72-1767 REV0 ENGINE - INTERMEDIATE CASE ASSEMBLY AND FITTINGS - HIGH PRESSURE (HP) INLET GUIDE VANES - INTRODUCTION OF AN ALTERNATIVE INTERMEDIATE PRESSURE (IP) AIR SEAL PANEL ASSEMBLY MADE FROM DIFFERENT MATERIAL
- SB72-A1653 REV1 HIGH PRESSURE (HP) COMPRESSOR VANES STAGES 4 TO 11 - INTRODUCTION OF A NEW SETTING PROCEDURE TO INCREASE THE MINIMUM VANE TIP CLEARANCE
- SB73-1604 REV1 ENGINE FUEL AND CONTROL - FUEL CONTROL SYSTEM - INTRODUCTION OF VARIABLE METERING ORIFICE (VMO) PUSH ROD WITH IMPROVED WEAR CHARACTERISTIC
- SB75-1518 REV0 AIR - AIRFLOW CONTROL REGULATOR AND ACTUATOR - INTRODUCTION OF REVISED ROLLING DIAPHRAGM - MOD.75-1518
- SB77-1305 REV0 ENGINE INDICATING - THERMOCOUPLES AND CABLES - INTRODUCTION OF INCREASED RANGE OF BALLAST RESISTORS

MODIFICATIONS FOUND EMBODIED:

- SB71-1147 POWER PLANT - VARIOUS - COLLECTOR AND EJECTOR SYSTEM TO PREVENT OIL ACCUMULATION IN BY-PASS DUCT
- SB72-1045 I.P. COMPRESSOR AIR SPLITTER - INTRODUCTION OF STRENGTHENED ANNULAR RING
- SB72-1202 COMPRESSOR INTERMEDIATE CASE - INTRODUCTION OF IMPROVED PROTECTIVE COATING ON CASING STRUTS
- SB72-1329 LP COMPRESSOR ROTATING FAIRING - NOSE CONE SPINNER WITH IMPROVED PROTECTIVE COATING
- SB72-1399 PART 1 - INTERMEDIATE CASE ASSEMBLY - FRONT FLANGE BORE - ADDITIONAL CORROSION PROTECTION
- SB72-1455 LP NOZZLE CASE AND VANES - INTRODUCTION OF LP TURBINE STAGE ONE NGV AND SEAL SUPPORT ASSEMBLY WITH HONEYCOMB SEAL
- SB73-1078 ENGINE - FUEL AND CONTROL - FUEL FLOW REGULATOR - INTRODUCTION OF AN APPROACH IDLE MECHANISM WITH INCREASED RANGE OF ADJUSTMENT



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CERTIFICAT de CONFORMITÉ

Table with 4 columns: Make - Constructeur (ROLLS-ROYCE), Model- Modèle (Tay 611-8 G-IV), Serial No. - No de Série (16844), Sales Order - Commande client (1365438). Includes TSN: 5803.2, CSN: 2302, TSO: NIL, CSO: NIL.

CONTINUED.....

- SB73-1079 ENGINE - FUEL AND CONTROL - FUEL FLOW REGULATOR WITH REDUCED THROTTLE VALVE LEVER BACKLASH
SB73-1131 FUEL AND CONTROL - FUEL FLOW REGULATOR - INTRODUCTION OF A TEST SCHEDULE AMENDMENT OF REDEFINE SETTING OF THE MINIMUM FLOW STOP
SB73-1166 ENGINE - FUEL AND CONTROL - FUEL FLOW REGULATOR - INTRODUCTION OF A SPEY TYPE LEVER HOUSING SEAL PLATE
SB73-1167 ENGINE - FUEL AND CONTROL - FUEL FLOW REGULATOR - INTRODUCTION OF IDLE RESET ADJUSTER MOUNTING BOLTS WITH WIRE LOCKING HOLES
SB73-1210 ENGINE - FUEL AND CONTROL - FUEL FLOW REGULATOR - INTRODUCTION OF A REDUCED VARIABLE METERING ORIFICE (VMO) PUSH ROD END FLOAT AND IMPROVEMENT TO THE VMO AND ACCELERATION CONTROL UNIT (ACU) PLUNGER AND SLEEVES
SB73-1211 ENGINE - FUEL AND CONTROL - ENGINE- FUEL AND CONTROL - FUEL FLOW REGULATOR - INTRODUCTION OF MODIFIED INPUT DRIVE SEAL RETAINER
SB73-1213 ENGINE - FUEL AND CONTROL - FUEL FLOW REGULATOR - INTRODUCTION OF A TWO PIECE SLIDER BLOCK AND PIN ASSEMBLY
SB73-1217 ENGINE - FUEL AND CONTROL - FUEL FLOW REGULATOR - INTRODUCTION OF REVISED DESIGN STANDARD OF REACH NUT
SB73-1219 ENGINE - FUEL AND CONTROL - FUEL FLOW REGULATOR - INTRODUCTION OF AN INCREASED CLEARANCE DASHPOT PISTON WITHIN THE SCHEDULING GOVERNOR
SB73-1280 FUEL AND CONTROL - SHUT-OFF COCK - INTRODUCTION OF A SHUT-OFF COCK WITH THE COLD START FACILITY DELETED
SB73-1281 ENGINE - FUEL AND CONTROL - FUEL FLOW REGULATOR - INTRODUCTION OF A FUEL FLOW REGULATOR WITH SOLUTIONS TO ELIMINATE CALIBRATION SHIFT
SB73-1301 FFR- INTRO OF AN FFR WITH REVISED ACU GOVERNOR WEIGHT ARM TO IMPROVE
SB73-1322 ENGINE - FUEL AND CONTROL - FUEL FLOW REGULATOR - INTRODUCTION OF A FUEL FLOW REGULATOR WITH A REVISED DOUBLE DATUM IDLE DESIGN
SB73-1592 ENGINE FUEL AND CONTROL - FUEL SYSTEM TUBES - INTRODUCTION OF A FUEL FLOWMETER TO HP FUEL PUMP TUBE ASSEMBLY WITH A NEW ROUTING AND INSTALLATION
SB76-1203 ENGINE CONTROLS - CONTROL LINKAGE - THROTTLE RELAY LEVER WITH RELIEF FOR P.L.A. TRANSDUCER
SB79-1411 OIL - OIL-TANK - INTRODUCTION OF REVISED OIL TANK ASSEMBLY WITH DELETED FUEL PIPE SUPPORT BRACKET
SB79-1426 OIL - ENGINE GASKETS - INTRODUCTION OF GASKET MATERIAL CHANGE TO KLINGERSIL C4430
SB76-1208 ENGINE CONTRO; - CONTROL LINKAGE - INTRUCTION OF THROTTLE RELAY LEVER MOUNTING BRACKET WITH INCREASED CLEARANCE FOR AIRCRAFT CONTROL ROD

NON-MODIFICATION SERVICE BULLETINS COMPLIED WITH:

- NMSB72-1496 REV0 H.P. COMPRESSOR STAGE 7 AND 12 AIR OFF-TAKE CLAMP INSPECTION
NMSB72-1603 REV1 LOW PRESSURE (LP) COMPRESSOR (FAN) BLADES - INSTRUCTION OF A REPAIR TO THE LP COMPRESSOR (FAN) BLADE LEADING EDGE PROFILE
NMSB72-1783 REV0 ENGINE - FAN BLADES - ULTRASONIC INSPECTION OF FAN BLADE ROOTS - NON-MODIFICATION SERVICE BULLETIN
NMSB72-1835 REV0 ENGINE -LOW PRESSURE (LP) COMPRESSOR (FAN) BLADES - DRY FILM LUBRICANT RE-APPLICATION CALCULATION METHOD - NON-MODIFICATION SERVICE BULLETIN
NMSB76-1319 REV0 ENGINE CONTROLS - ADJUSTMENT OF THE SPINDLE RETAINING PLATE ON THE THROTTLE RELAY LEVER BRACKET TO PREVENT OSCILLATION OF THE SPINDLE



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TECHNICAL INSTRUCTION COMPLIED WITH:

CTTI-122/16 POWER PLANT - LIST OF PARTS TO BE OVERHAULED FOR ENGINES PROCESSED AT RRC FOR MID LIFE OR OVERHAUL

TECHNICAL VARIANCES EMBODIED:

- List of technical variances including TV122930, TV185521, TV187937, TV188847, TV190119, TV191667, TV191900, TV191934, TV192095, TV192275, TV192612, TV192832, TV192837, TV192846, TV193100, TV193113, TV193114, TV193169, TV193316, TV193355, TV193358, TV193360, TVH19427, TVH21758.

REPAIR SCHEMES EMBODIED:

- List of repair schemes including HRS3101, HRS3102, HRS3122, HRS3135, HRS3164, HRS3165, HRS3166, HRS3169, HRS3170, HRS3183, HRS5035, HRS5038, HRS5040, HRS5663, HRS5711.



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TSN: 5803.2		TSO: NIL	
CSN: 2302		CSO: NIL	

REPAIR INSTRUCTIONS EMBODIED:

- TARI5029 WELD FRETTED STOP LUGS
- TARI5038 REPAIR CRACKS AT 4-OFF AXIAL SEAMS LOCATION (INNER SURFACE)
- TARI5076 REPAIR DETACHED SOLEPLATE (PREVIOUSLY TARI 1007)
- TARI5095 GRIND TO OBTAIN BLU BED CHECK
- TARI5113 REPAIR CRACK ON THE AIRFOIL, LEADING EDGE, INNER AND OUTER PLATFORMS BY WELDING.
PART 1: REPAIR THE AIRFOIL, LEADING EDGE, INNER AND OUTER PLATFORMS
PART 2: PRE AND POST HEAT TREATMENT FOR WELD REPAIR
- TARI5040 COOLING AIR TRANSFER TUBE - WELD FRETTED ANTI-ROTATION PIN

AIRWORTHINESS DIRECTIVES

SEE ATTACHED AIRWORTHINESS DIRECTIVES / SERVICE BULLETIN STATUS SHEET.

NOTES ABOUT THIS PIECE OF EQUIPMENT:

THE ENGINE WAS RETURNED FROM SERVICE DUE TO " OVERHAUL" AND WAS OVERHAULED, MODIFIED AND TESTED IN ACCORDANCE WITH E-TAY-1RR REV. 108, ROLLS-ROYCE NMSB70-1050 REV. 9 APPENDIX 6 AND CHAPTER 5 OF THE MAINTENANCE MANUAL M-TAY-1RR REV.135 AND CUSTOMER P.O. 9000617083.

ENGINE REAR LINKS (P/N: JR30345 S/N: BSEXY0109 & P/N: JR30346 S/N: BSEXY0108) AND EXHAUST COLLECTOR SUPPORT STRUTS WERE OVERHAULED IN ACCORDANCE WITH ROLLS-ROYCE ENGINE MANUAL REF: E-TAY-1RR REV.108. FRONT MOUNT (P/N: JR28161 S/N: RT1957) IS NEW.

MAINTENANCE OPERATIONS 4, 6 TO 23 OF THE SCHEDULED MAINTENANCE CHECKS IN CHAPTER 5 OF THE MAINTENANCE MANUAL HAVE BEEN COMPLIED WITH. CHIP DETECTORS WERE REMOVED, CLEANED AND RE-INSTALLED AFTER TEST.

-88 H.P.1 TURBINE BLADES P/N: JR34248 REFITTED (TSN: 5803.2)

- SB71-1100 AND SB76-1130 MUST BE COMPLIED WITH PRIOR TO ENGINE INSTALLATION AS DETAILED IN EACH ROLLS-ROYCE SERVICE BULLETIN.

- SB71-1147 FOUND EMBODIED IN PART, ALL RELEVANT PARTS MUST BE COMPLIED WITH PRIOR TO ENGINE INSTALLATION AS DETAILED IN ROLLS-ROYCE SERVICE BULLETIN 71-1147.

- SB73-1354 TO BE COMPLIED WITH DURING ENGINE GROUND RUNS AS DETAILED IN THE ROLLS-ROYCE SERVICE BULLETIN. THIS SB MUST BE COMPLIED WITH ON BOTH ENGINES INSTALLED.

- HPC STAGE 10 ROTOR DISC: PART NUMBER JR18750 SERIAL NUMBER DETM17866 IDENTIFY VIBROPEEN THE HPC STAGE 10 ROTOR DISC THE NEXT POSSIBLE SHOP VISIT I.A.W. TV193355 IF LEVEL OF STRIP PERMITS ACCESS.

- HPC STAGE 11 ROTOR DISC: PART NUMBER JR18751 SERIAL NUMBER DETM18127 IDENTIFY VIBROPEEN THE HPC STAGE 11 ROTOR DISC THE NEXT POSSIBLE SHOP VISIT I.A.W. TV193316 IF LEVEL OF STRIP PERMITS ACCESS.



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Make - Constructeur ROLLS-ROYCE	Model- Modèle Tay 611-8 G-IV	Serial No. - No de Série: 16844	Sales Order - Commande client 1365438
TSN: 5803.2	TSO: NIL		
CSN: 2302	CSO: NIL		

MODULE:	S/N:	TSN / CSN	WORK / STATUS:	SALES ORDER #:
M01100AB	TM2745	5803.2 / 2302	OVERHAULED	1365438
M02100AA	TM2744	5803.2 / 2302	OVERHAULED	1365438
M03100AC	TM2744	5803.2 / 2302	OVERHAULED	1365438
M04100AA	TM2743	5803.2 / 2302	OVERHAULED	1365438
M05100AA	TM2743	5803.2 / 2302	OVERHAULED	1365438
M06100AA	TM2744	5803.2 / 2302	OVERHAULED	1365438
M07100AB	TM2744	5803.2 / 2302	OVERHAULED	1365438

CERTIFICATION: CONFORMITY STATEMENT

Certified that the whole of the material and/or parts and/or engine detailed above (except As otherwise specified) conforms to all specifications relevant thereto in accordance with the Conditions of the contract or purchase order.

CERTIFICATION: DÉCLARATION DE CONFORMITÉ

Certifie que tous les matériaux et/ou pièce et/ou le moteur détaillés ci-haut (sauf si Indiqué autrement) sont conformes aux spécifications pertinentes et ce faisant aux conditions du contrat ou du bon de commande.

Authorized signatory - Signataire autorisé

THOMAS PAGEL

Stamp - Estampille



Date

NOVEMBER 20, 2018

AMO 1-58



Rolls-Royce

Canada Limited

LIFE LIMITED PARTS LOG SHEET

OPERATOR: ALBINATI AERONAUTICS SA
 ENGINE S/N: 16844
 ENGINE MARK: TAY 611-8
 ENGINE TSN: 5803.2
 ENGINE CSN: 2302
 S/O 1364991

COMPONENT	SECTION ATA	PART NUMBER	SERIAL NUMBER	TOTAL HRS	TOTAL CYCLES	FLT. LIFE (CYC)	CYCLES REMAIN	REMARKS
LP COMPRESSOR DISC	72-31-11, 01-100	JR34563A	FTTDK00364	5803.2	2302	11600	9298	
DISC IPC STAGE 1	72-33-31, 01-025	JR28921	CTDK5648	5803.2	2302	20000	17698	
DISC IPC STAGE 2	72-33-32, 02-025	JR28922	CTDK5617	5803.2	2302	20000	17698	
DISC IPC STAGE 3	72-33-33, 03-025	JR29568	STDK20065	5803.2	2302	15000	12698	
IP/LP COMP DRIVE SHAFT	72-33-40, 04-100	JR35036	PAVN5360	5803.2	2302	11500	9198	
HP COMP REAR SHAFT & SLEEVE	72-37-31, 01-005	JR35028A	PABX20306/A	5803.2	2302	20000	17698	
LP COMP DRIVE SHAFT	72-37-32, 02-085	JR35037	SEABX1460	5803.2	2302	10600	8298	
DISC HP COMP STAGE 1	72-37-33, 03-165	JR18049	RRSUNDC94L	0.0	0	9850	9850	New material
DISC HP COMP STAGE 2	72-37-33, 03-250	JR18742	SETP09126	5803.2	2302	20000	17698	
DISC HP COMP STAGE 3	72-37-33, 03-295	JR18743	EEBE8607	0.0	0	20000	20000	New material
DISC HP COMP STAGE 4	72-37-34, 04-340	JR18744	DETP9890	5803.2	2302	20000	17698	
DISC HP COMP STAGE 5	72-37-34, 04-385	JR18746	668CCR	9690.4	4396	20000	15604	EX:16291
DISC HP COMP STAGE 6	72-37-34, 04-410	JR18748	SETP09031	5803.2	2302	20000	17698	
DISC HP COMP STAGE 7	72-37-34, 04-435	JR13814	SETP09117	5803.2	2302	20000	17698	
DISC HP COMP STAGE 8	72-37-35, 05-460	JR13815	DETM18108	5803.2	2302	20000	17698	
DISC HP COMP STAGE 9	72-37-35, 05-485	JR13816	EEBH12957	0.0	0	20000	20000	New material
DISC HP COMP STAGE 10	72-37-35, 05-510	JR18750	DETM17866	5803.2	2302	20000	17698	
DISC HP COMP STAGE 11	72-37-35, 05-535	JR18751	DETM18127	5803.2	2302	20000	17698	
DISC HP COMP STAGE 12	72-37-36, 06-100	JR18449	SETM13422	9682.6	4586	9350	4764	*, EX:18009
HP TURB SHAFT & SEAL ASSY	72-41-31, 01-010	JR35030A	PABX20153/A	5803.2	2302	12000	9698	
DISC HP TURB STAGE 1	72-41-32, 02-010	JR28878	SETM09937	5803.2	2302	10000	7698	
DISC HP TURB STAGE 2	72-41-33, 03-250	JR57978	RRSUNDD76R	0.0	0	10000	10000	New material
DISC HP TURB STAGE 2 SLEEVE	72-41-33, 03-260	NPH7674	200141D09344	0.0	0	12850	12850	New material
LP TURB SHAFT & SLEEVE ASSY	72-52-21, 01-450	JR35039	PAVN6050	9682.6	4586	11500	6914	EX:18010
DISC LP TURB STAGE 1	72-52-22, 02-400	JR32318A	DETM14707/A	7578.7	2896	10000	7104	EX:16701
DISC LP TURB STAGE 2	72-52-23, 03-440	JR57973	RRSUNDC18N	0.0	0	10000	10000	New material
DISC LP TURB STAGE 3	72-52-24, 04-400	JR32320A	DETM19641	6988.8	3976	10000	6024	EX:16920

Items identified thus "*" Indicate Limiting Part (s)

Note: The declared flight life and cycles remaining information quoted is correct at date of issue, but will not be updated by Rolls-Royce Canada Limited

Items identified thus "#" were not stripped; Data provided by operator.

J. Sule



Rolls Royce (Canada) Limited

Date: 19-Nov-2018